



URANUS
VOYAGER 2

Helium-3 Mining Aerostats in the Atmospheres of the Outer Planets

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Reference: Van Cleve et. al. "Helium-3 Mining Aerostats in the Atmospheres of the Outer Planets", 2002

Art by David Seal of JPL



Imagine an Interplanetary Future

Where -



- d-He3 fusion produces most of Earth's energy needs without radioactivity or carbon emissions
- Space transportation has been revolutionized by an efficient fusion propulsion system with exhaust velocity up to 0.088 c
- Space commerce is stimulated by the existence of an interplanetary cargo worth \$3-M a kilogram
- Unmanned probes travel to the nearest star systems with flight times less than a human lifetime



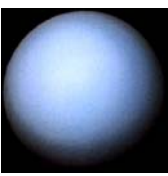
He-3 Fusion for Energy & Propulsion



- reactants are stable and storable
- products are energetic, charged and stable
 - Efficient electrical generation from MHD
 - No activation and embrittlement of reactor vessel
 - Efficient conversion to thrust with exhaust velocity up to 0.088 c --> ~50 yr interstellar flight using known physics.
- 3.6×10^{14} J/kg of d-He3 mixture = 1.0×10^8 kWh/kg
 - Fuel is about 20% of the kWh cost of electricity
 - If electricity is 15¢/kWh then He3 has a value of \$3M/kg

He-3 is one of the few commodities worth interplanetary freight costs

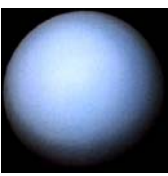
Why Outer Planets for He-3?



- Earth: breeding of tritium from either isotope of lithium by neutron bombardment, tritons decay to He-3.
 - Containment, waste problems same as d-t fission.
 - USA has no current capability.
 - Lithium inventory?
- Moon: solar wind implanted in regolith, 10 ppb (10^{-8}) by mass in uppermost few meters. ~1000 yr of 2001 energy needs- a starter catalyst?.
- Outer planets: primordial He3, ~10 parts per million (10^{-5}), ~ 10^9 yr of 2001 energy needs- the ultimate energy source?.



Which Outer Planet-Jupiter



Pro:

- Closest to Earth and Sun

Con:

- Huge gravity means return vehicle has mass ratio >20 (nuclear thermal $I_{sp} = 900 \text{ s}$)
 - No mass budget left for cargo!
- A lot hotter at a any given density
 - Galileo probe killed by heat not by pressure



Which Outer Planet-Saturn



Pro:

- Not as far as Uranus and Neptune
- Rapid rotation substantially reduces ΔV to orbit

Con:

- Seen as depleted ~5x in Helium compared to other outer planets
 - reanalysis of Voyager data 20 yr later restores that 5x- maybe
 - won't know for sure until we send an entry probe
- Rings as a navigation hazard
 - need close-in, co-orbiting mission to look

Which Outer Planet-Uranus



Pro

- Primordial He3 abundance ?
- ΔV to orbit requires mass ratio < 5
- Closer than Neptune

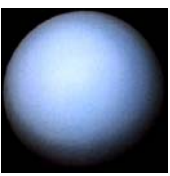
Con

- Axial tilt complicates interplanetary travel
- Twice as far from Earth as Saturn

Uranus may be the closest planet without major possible problems -- but we must return to both Saturn to be sure



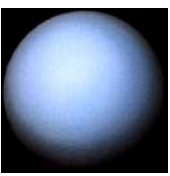
Do we really know how much He3 is there?



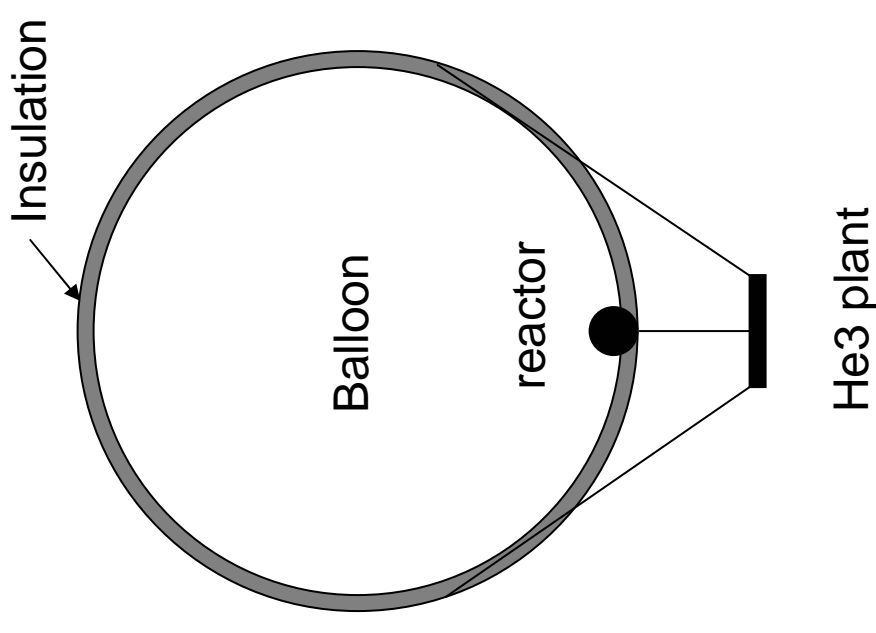
- He3/He4 cannot be measured by remote sensing
- He3/H₂ and He3/He4 ratios have been measured *in situ* only by *Galileo* at Jupiter
- He3/He4 ratio of 10^{-4} to 1.5×10^{-4} from meteors, solar wind, cosmology
- Use *Galileo* results for He3/He4 = 10^{-4} and Voyager results (?) for He4/H₂

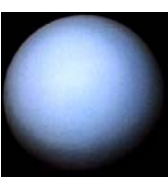


He-3 Mining with Balloons



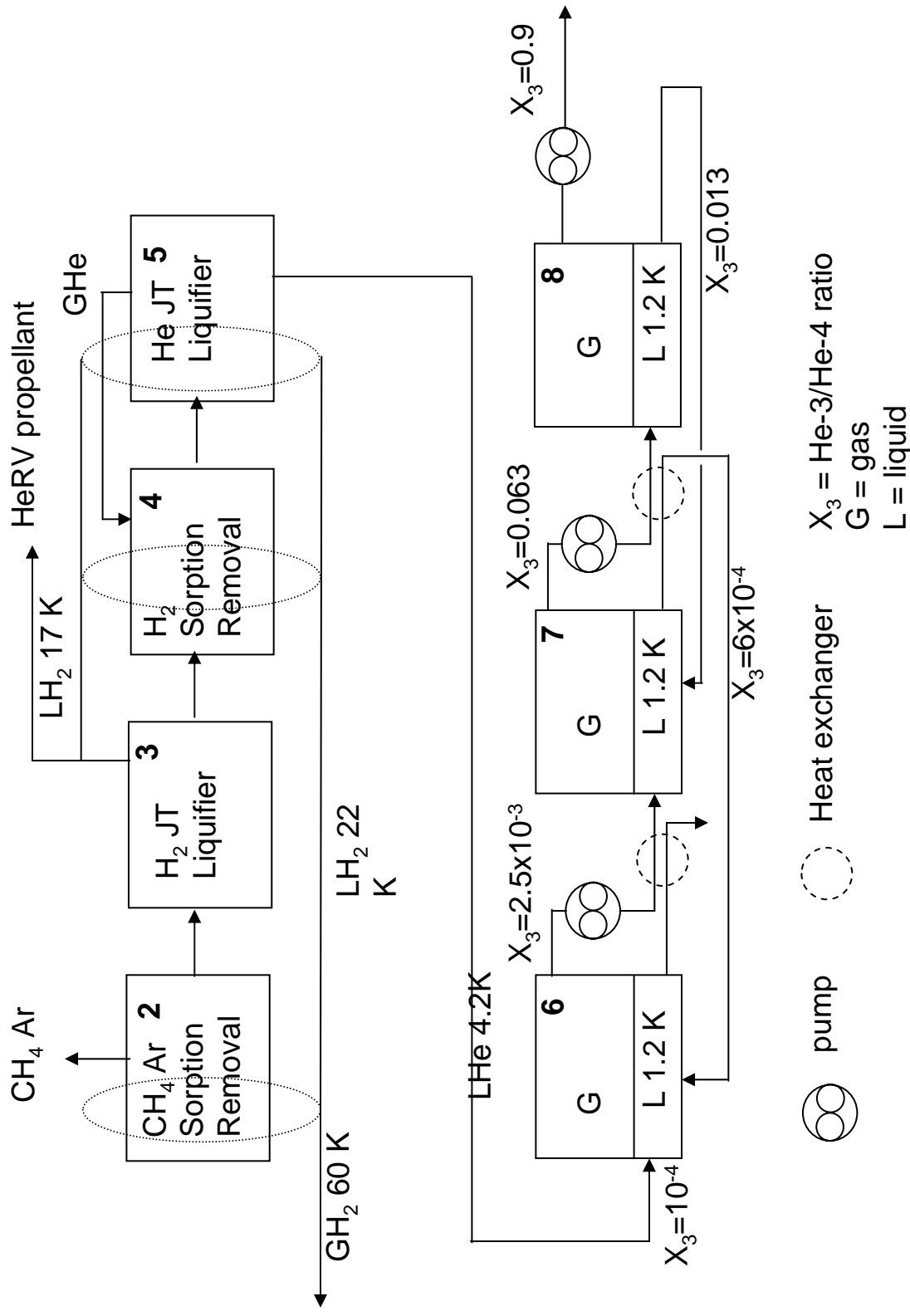
- Balloon diameter : 80 m
- Total Plant mass: 146 tonnes
- Return vehicle: 59 tonnes
- Total lift needed: 205 tonnes





Notional Distillation Plant Concept

Thinking Big about our Space Cryogenics Future





Energy Economics

$\text{He3}/\text{H}_2 = 10 \text{ ppm}$



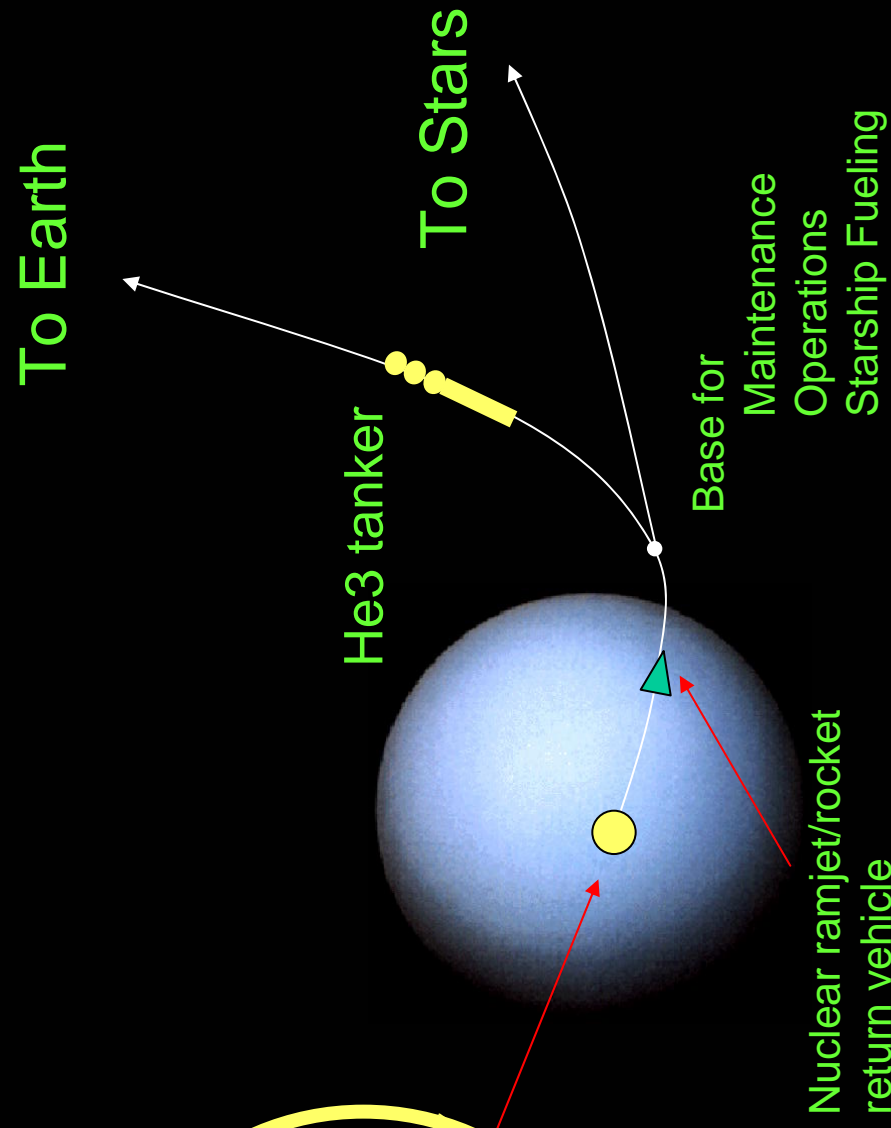
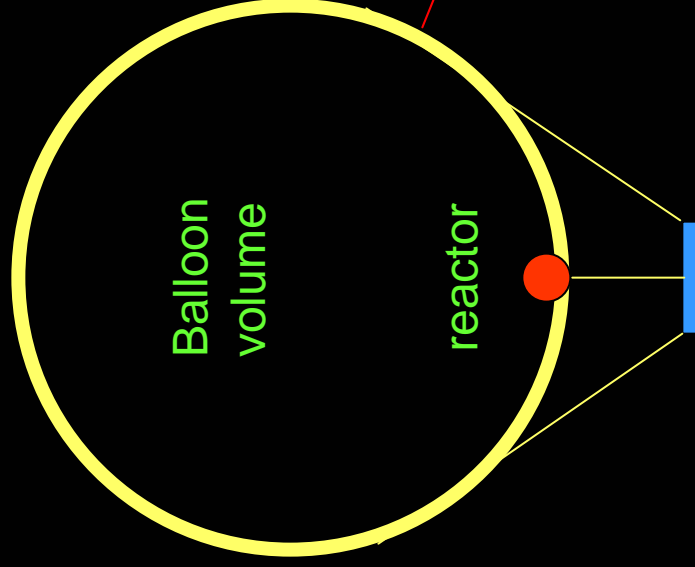
Table 1

Stage	Process	Energy (J)/g He3
1, 2, 3	cool atmosphere to 16 K	7.2×10^7
3	liquify H_2 at 16 K	3.2×10^8
5	cool He from 16 K to 4.2 K	1.3×10^7
5	liquify He at 4.2 K	1.1×10^7
6	cool LHe from 4.2 to 1.2 K	1.2×10^7
total		4.3×10^8

Transportation on 2 yr trajectory: $5 \times 10^7 \text{ J/g He3}$
Energy released: $6 \times 10^{11} \text{ J/g He3}$
Theoretical energy payback: ~1000

The Persian Gulf of the Solar System, 2150

Mining aerostat



The most valuable interplanetary commodities are refined He-3, deuterium, and heavy metals



Next Steps

- Jupiter Icy Moons Orbiter (JIMO)
 - nuclear fission-powered
 - electric propulsion flight system
 - Big deal: 20 tonnes, >\$4 B, 10 kWe
 - First of a series: Project Prometheus
- Saturn Ring Observer
- Uranus/Neptune Orbiter with Probes
- Self-deploying balloon probes for Mars, Titan
- Discovery/New Frontiers missions to other resource sites (Moon, asteroids, comets) for interplanetary commodity economy





A Trial Balloon ?



Scientific balloon missions to outer planets, using Pu RTGs and/or O₂ burners, to study

- He3/He4 and He/H₂ ratio
- pressure vs. temperature for $1 < p < 100$ bar
- trace gas composition
- entry, deployment, and telemetry engineering experiments

A science balloon could be as small as 2.8 m diameter, and use at most 7 kg of Plutonium as a heat and power source